

# Analysing the efficiency of thermopressor application in the charge air cooling system of combustion engine

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**Abstract.** As the analysis of the research results has shown, the use of a thermopressor makes it possible to increase the fuel and energy efficiency of a ship power plant in a wide range of the operation parameters. It can be achieved by cooling the charge air before the engine inlet receiver and by reducing the compression work of the turbocharger. A scheme with the thermopressor application in the cooling system of a low-speed main engine and in the system for utilizing the exhaust gases heat in a heat recovery boiler of one and two pressures was proposed. The use of thermopressors led to a decrease in the compressor power consumption, and therefore in the turbine required power. Therefore, it was appropriate to pass (bypass) the excess amount of gas past the turbine. Accordingly, the thermal potential of exhaust gases was increased. As a result, the temperature of gases at the inlet to the heat recovery boiler was increased by 10-15 °C, and gases heat was increased by 10-15% respectively. The obtained additional steam is advisable to use for driving the utilization turbine generator, thereby reducing the load on the ship's power plant, with a corresponding decrease in fuel consumption of diesel generators by 2-4%.

## 1 Introduction

Increasing the power of ship power plants based on internal combustion engines was carried out by increasing the flow rate of the working fluid (charge air) and the useful work of combustion products expanding. At the same time, there was a simultaneous reduction in the work for air compressing, which was achieved by cooling air and deep utilization of the combustion products energy. That, in turn, led to a decrease in specific fuel consumption and an increase in efficiency.

## 2 Literature review

Low-, medium- and high-speed internal combustion engines are widely used in power plants of transport and stationary energy. The air parameters at the turbocharger discharge and at the inlet to the cylinders significantly affect the fuel and energy efficiency of the engine. Every 10 degrees an increase in air temperature causes a decrease in the effective efficiency  $\eta_e$  of marine low-speed diesel engines by 0.5–0.7% [1, 2] with a corresponding increase in specific fuel consumption  $g_e$  [3]. The decrease in engine power is caused by a decrease in the mass air flow in the cylinders due to a decrease in the air density with an increase in temperature [4, 5].

The development of modern energy efficient technologies for power plants can cause a low level of waste heat use [6, 7]: the temperature of gases after the heat recovery boilers is about 180 °C; charge air

temperature is 140–220 °C; the temperature of the engine cylinders cooling water is 90-120 °C. Heat with such a thermal potential can be used in several ways:

1. The use of the combustion products heat for the steam production in the heat recovery boiler (reduction of the specific fuel consumption  $g_e$  is 2–3%) [8, 9].
2. Turning to heat recovery boilers with two-pressure circuits (additionally increases the capacity of the utilization turbine generator by 15–30%) [10, 11].
3. To improve the turbocharge system [12, 13].
4. The use of the high-temperature cooling method of the internal combustion engine [14].
5. The use of combined energy production technologies with simultaneous production of heat (steam, hot water), electrical energy (generator, shaft generator) and cold (heat-using ejector and absorption refrigeration machines) [15].

The most common way to improve the fuel and energy efficiency of a power plant is contact cooling of the air flow by water injection [16, 17]. It is promising to cool the charge air of the internal combustion engine with a thermopressor which provides an increase in efficiency and a reduction in fuel consumption due to a decrease in temperature and an increase in compressed air pressure. It leads, in turn, to a decrease in the compression power consumption [18]. Changing operating conditions of power plants, and, consequently, the thermal load on the thermopressor, require rational organization of work processes and the determination of a rational design pressure increase that would provide the maximum effect.

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Thermopressor technologies are based on the implementation of the thermogasdynamic compression effect in special contact heat exchangers (aerothermopressors or thermopressors), which consists in increasing the pressure while decreasing the temperature during the evaporation of a finely dispersed liquid injected into a gas flow moving at a speed close to sound [19, 20].

On the basis of the presented analysis, on the current state of development of technologies for increasing the fuel and energy efficiency of power plants based on internal combustion engines, the main task of the research was formulated: to develop and analyze improved schemes of thermopressor systems for utilizing the energy of combustion products with cooling the working fluid (air) of the engines, principles and methods of their implementation.

### 3 Research methodology

A comparative-calculation method of research was used in this work to determine the thermodynamic and energy efficiency of a thermopressor cooling system as part of a power plant. A proprietary software package was developed to use this method and it allows: to simulate working processes in a thermopressor; to calculate the main structural elements of the thermopressor; to calculate the energy efficiency and main indicators of the engine when using thermopressor cooling systems, taking into account changes in climatic and hydrometeorological conditions, as well as particular operating modes of the power plant.

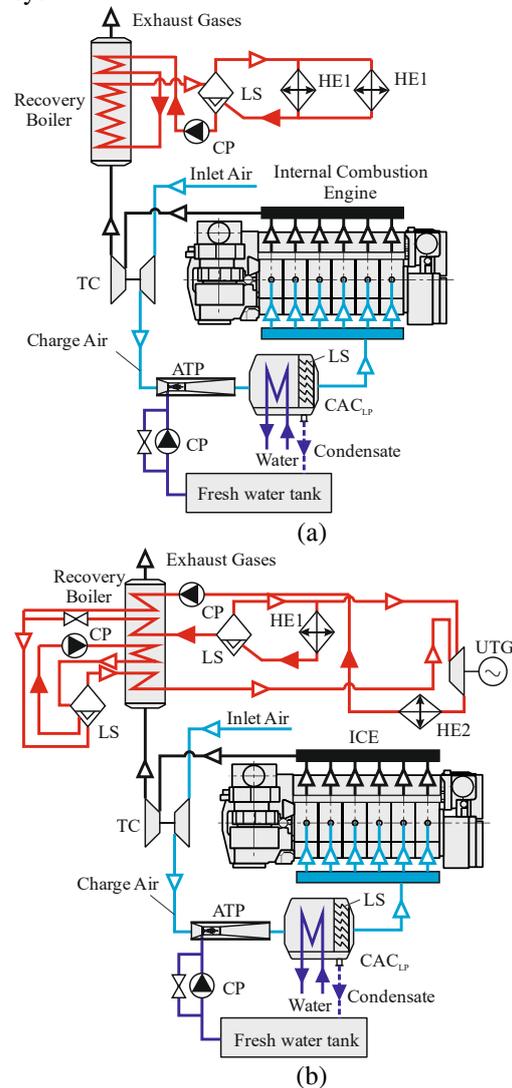
The calculation of the thermopressor characteristics (air pressure  $P_{air}$ , air temperature  $T_{air}$ , air velocity  $w_{air}$ , water velocity  $w_w$ , etc.) and its basic geometric parameters of the flow path under specified operating conditions of the power plant engine was carried out according to the indicated methods [21-23]. The calculation took into account the pressure loss in the nozzle (confuser), working chamber, diffuser [24, 25], as well as with the frontal resistance of injected water droplets [26, 27].

The calculation of the main parameters of the internal combustion engine was carried out using the software packages Diesel-RK and CEAS (MAN B&W) [28] taking into account the change in the air parameters at the inlet to the engine cylinders and partial operating modes. To use these software systems, an additional program was developed in which the parameters of wet gas (air) were determined at the inlet and outlet of the turbocharger.

### 4 Results

A scheme in which the thermopressor is used as a charge air cooler behind a turbocharger of low-speed internal combustion engines with a waste gas heat recovery system in a waste heat recovery boiler is shown in Fig. 1a. The analysis was carried out in relation to the standard traditional scheme for charge air cooling of ship internal combustion engines. The calculations of a low-speed main engine 5S50MC-C (MAN B&W) with

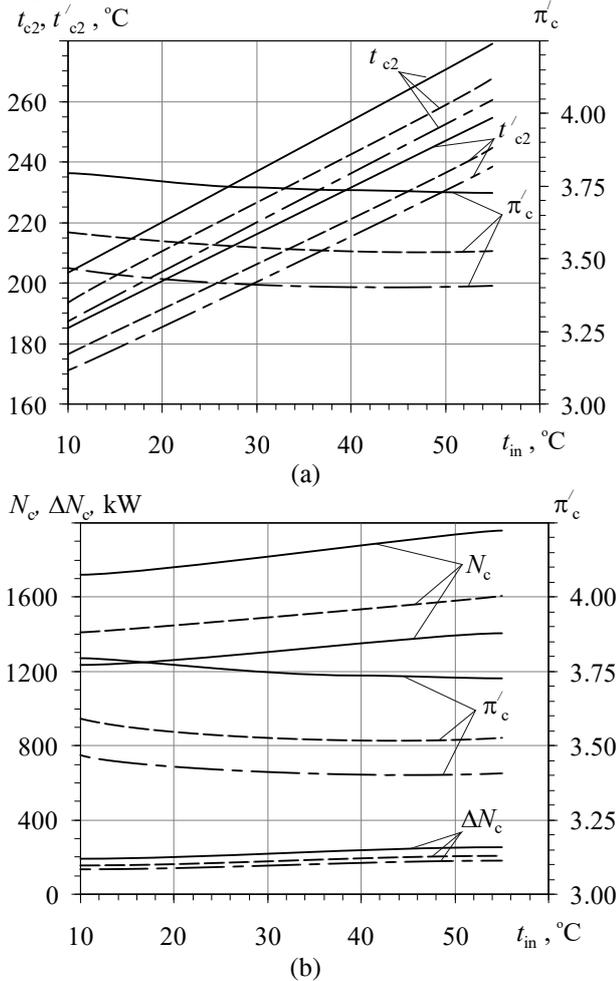
a power of  $N_b = 8300$  kW and a speed of  $n = 105$  rpm were carried out. The calculation of the thermopressor parameters was carried out taking into account its joint work with a turbocharger of the pressure-charging system.



**Fig. 1.** A scheme of the charge air cooling system with a thermopressor for low-speed marine engines (a); a scheme of a thermopressor charge air cooling system with integrated waste gas heat recovery in the heat recovery boiler and UTG (b): LS – liquid separator; CAC – charge air cooler; TC – turbocharger; ATP –thermopressor.

The increase in pressure in the outlet thermopressor  $\Delta P_{atp}$  significantly depends on the value of the temperature decrease during cooling  $\Delta t_{atp}$ , and therefore the air temperature at the inlet  $t_{atp1}$  is very important. The temperature in front of the thermopressor corresponds to the discharge air temperature of the turbocharger. In the scheme with one-stage compression, the temperature in front of the thermopressor is  $t_{c2} = 190-270$  °C (Fig. 2a). The temperature behind the turbocharger is the higher, the higher the inlet temperature and the degree of pressure increase in the turbocharger  $\pi_c$ . Since the thermopressor cooling system, in fact, is a two-stage air compression (the first stage is a turbocharger, the second stage is a thermopressor), the compression ratio  $\pi_c$  of such a scheme will be lower (Fig. 2b), and therefore  $t_{c2} = t_{c2}' =$

170–250 °C, that is, the temperature in front of the thermopressor decreases, and accordingly the compression ratio in the air thermopressor  $\pi_{atp}$  will also be lower. Since the required compression ratio for the turbocharger  $\pi_c$  decreases, the compressor work for compression  $l_c$  decreases accordingly. For a system with total  $\pi_{c.atp} = 4.25$  the decrease in work is  $\Delta l_c = 10–13$  kJ/kg; with total  $\pi_{c.atp} = 4.00 - \Delta l_c = 9.0–12.5$  kJ/kg; with total  $\pi_{c.atp} = 3.85 - \Delta l_c = 8.5–12.0$  kJ/kg (Fig. 3). Reducing the work on compression  $l_c$  allows to reduce the compressor power (Fig. 2b) by  $\Delta N_c = 100–200$  kW (10.0–11.5 %), with the same air consumption in the engine.



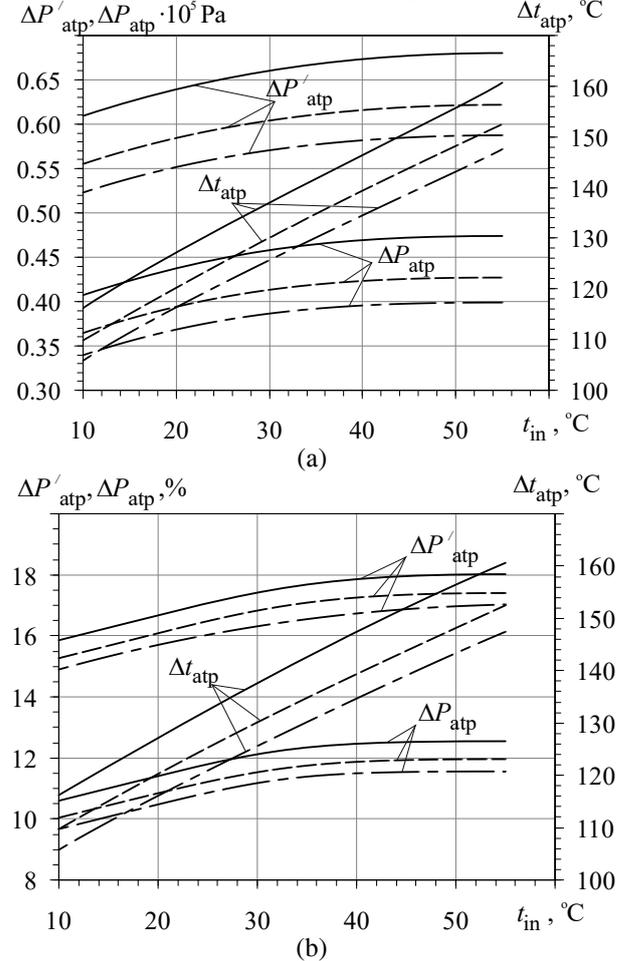
**Fig. 2.** Dependences of the air temperature at the turbocharger discharge without a thermopressor  $t_{c2}$ , by using a thermopressor  $t'_{c2}$ , degree of pressure increase  $\pi_c'$  (a) turbocharger power  $N_c$ , reducing the turbocharger power  $\Delta N_c$  at different total pressure increase  $\pi_{c.atp}$  on the inlet air temperature  $t_{in}$ :

————— -  $\pi_{c.atp} = 4.25$ ; - - - - -  $\pi_{c.atp} = 4.00$ ;  
 - · - · -  $\pi_{c.atp} = 3.85$ .

The minimum temperature at the outlet of the thermopressor  $t_{atp2}$  was taken to be a temperature 2–3 °C higher than the dew point temperature. It was taken into account that water injection continues in the thermopressor until the air is completely saturated, that is,  $\phi = 100\%$ . The decrease in the air temperature is  $\Delta t_{atp} = 110–160$  °C, which makes it possible to increase the air pressure after the turbocharger by  $\Delta P'_{atp} = 520–670$  kPa (Fig. 3a) with "ideal" compression (without taking

into account friction losses about the channel wall), and with real compression in the thermopressor air pressure increase is  $\Delta P_{atp} = 340–480$  kPa, this corresponds to  $\Delta P'_{atp} = 15–18$  % and  $\Delta P_{atp} = 10–13\%$  (Fig. 3b).

At the same time, the degree of decrease in air temperature in the thermopressor was  $T_{atp1}/T_{atp2} = 1.30–1.45$ . The thermopressor is the second stage of compression in the pressure-charging system; accordingly, it is appropriate to evaluate its efficiency by the degree of pressure increase  $\pi_{atp} = 1.10–1.13$ .



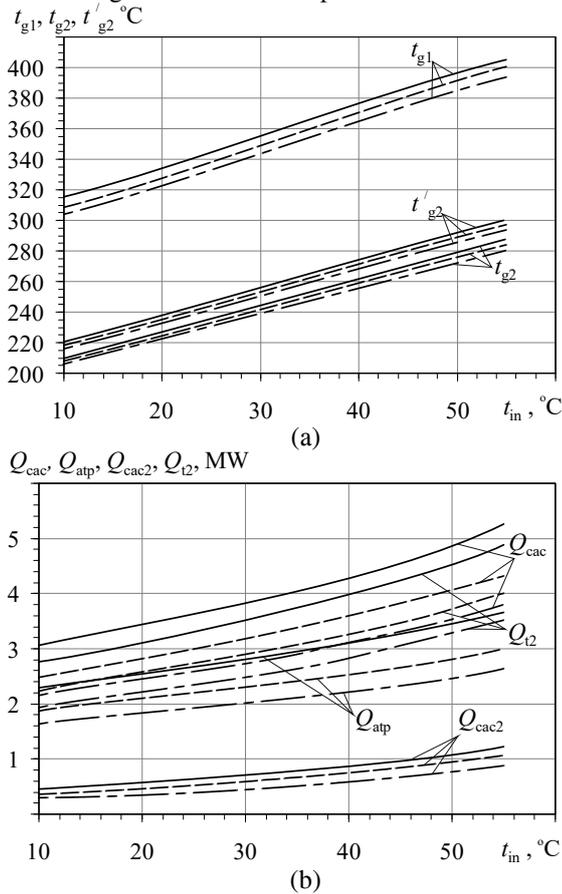
**Fig. 3.** Dependences of the decrease in the thermopressor air temperature  $\Delta t_{atp}$ , an increase in the air pressure at the discharge with "ideal" compression  $\Delta P'_{atp}$ , with real compression  $\Delta P_{atp}$  (a, b) at different total pressure increase  $\pi_{c.atp}$  on the inlet air temperature  $t_{in}$ :

————— -  $\pi_{c.atp} = 4.25$ ; - - - - -  $\pi_{c.atp} = 4.00$ ;  
 - · - · -  $\pi_{c.atp} = 3.85$ .

A decrease in air temperature and an increase in pressure in the thermopressor are quite significant, which affects the decrease in flow rates in the turbocharger, as a result, the exhaust gases temperature of the utilization turbine (UT) increases (Fig. 4a). Thus, at a constant temperature of gases at the UT inlet ( $t_{g1} = 300–400$  °C), the temperature of the exhaust gases is  $t_{g2} = 205–285$  °C, in the scheme by using the thermopressor the temperature of the exhaust gases is  $t_{g2} = 220–300$  °C, which is 15 °C higher for the basic version.

Comparison of the thermopressor with the heat exchanger, from the point of view of the heat exchanger (charge air cooler), shows that the heat load on the thermopressor is  $Q_{atp} = 1600–3700$  kW (Fig. 4b).

However, the air temperature at the discharge of the thermopressor is still high and amounts to  $t_{atp} = 65\text{--}85$  °C. Therefore, it is advisable to install an additional heat exchanger behind the thermopressor. The heat load on such charge air cooler is  $Q_{cac2} = 200\text{--}1200$  kW, hence the total heat load on the thermopressor and the additional heat exchanger is  $Q_{t2} = 2000\text{--}4800$  kW, which is less than for the standard charge air cooling system  $Q_{st2} = 2400\text{--}5200$  kW (7–15 %). The increase in the thermal load on the standard charge air cooler can be explained by reducing the air temperature in the turbocharger with the thermopressor at 20–25 °C.



**Fig. 4.** Dependences of the exhaust gases temperature before UT  $t_{g1}$ , exhaust gases temperature UT (for a circuit without a thermopressor)  $t_{g2}$ , exhaust gases temperature UT (for a circuit with a thermopressor)  $t'_{g2}$  (a), heat load on: heat exchanger  $Q_{cac}$ , for thermopressor  $Q_{atp}$ , for additional heat exchanger  $Q_{cac2}$ , total heat load  $Q_{t2}$  (b) at different total pressure increase  $\pi_{c.atp}$  on the inlet air temperature  $t_{in}$ :

—————  $\pi_{c.atp} = 4.25$ ; - - - - -  $\pi_{c.atp} = 4.00$ ;  
 - · - · -  $\pi_{c.atp} = 3.85$ .

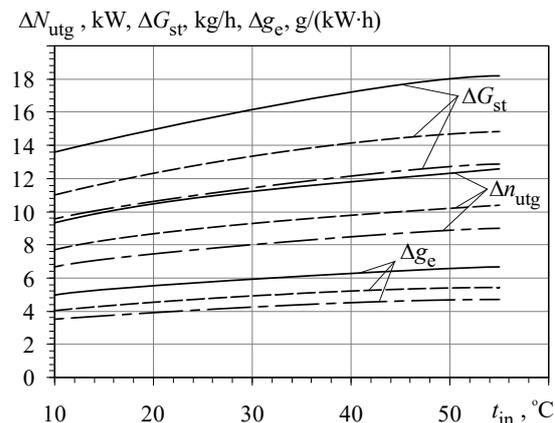
The efficiency of thermopressor operation was estimated taking into account the energy consumption for the supply of fresh water for injection in the thermopressor nozzle. According to calculations, the water consumption for injection into the thermopressor is  $G_w = 0.6\text{--}1.2$  kg/s, while the pump drive power is  $N_w = 0.3\text{--}0.7$  kW. The moisture content of the air increases by  $\Delta d_{atp} = 40\text{--}60$  g/kg. The injection pump power is rather small (0.3–0.5%) compared to the decrease in the compressor power  $\Delta N_c = 100\text{--}200$  kW.

The use of the thermopressor leads to a decrease in the turbocharger power, and then the unnecessary heat

drop (work) of the TC turbine decreases, the required power of the turbine and the required gas flow rate decrease. Hence, it is appropriate to pass the excess amount of gas past the turbine, due to which the exhaust gases temperature in front of the heat recovery boiler increases. Hence, the thermal potential of waste gases increases, which can be used in a waste heat recovery boiler. So the temperature at the heat recovery boiler inlet grows by almost  $\Delta t_{rb} = 10\text{--}15$  °C, taking into account the temperature of the gases at the heat recovery boiler outlet is constant and equal to  $t_{rb2} = 160$  °C, the additional heat load (with the corresponding additional consumption steam) is  $\Delta Q_{rb} = 150\text{--}300$  kW (10–15%).

As shown by the calculations of the thermal scheme, the amount of steam generated in the heat recovery boiler is more than enough to meet the needs of this dry cargo ship. So, it is advisable to use the obtained additional steam to drive a utilization turbine generator (UTG), thereby reducing the load on the ship's power plant, with a corresponding decrease in fuel consumption for diesel generators.

A scheme of the thermopressor charge air cooling system with integrated heat recovery from exhaust gases in the heat recovery boiler with two pressure circuits and UTG is shown in Fig. 1b. Analysis of the UTG operation shows (Fig. 5) that an increase in heat power in a waste heat recovery boiler of two pressures allows an increase in steam production by  $\Delta D_{utg} = 0.07\text{--}0.12$  kg/s (250–430 kg/h), which increases the power of UTG by  $\Delta N_{utg} = 45\text{--}90$  kW (6–10% of the rated power of the UTG). When two MAN B&W 7L16/24 diesel generators are installed on board with a rated power  $N_e = 770$  kW and a specific fuel consumption  $g_e = 195$  g/(kW·h), a decrease in the load on the ship's power plant when the UTG is turned on (Fig. 5) reduces the specific fuel



**Fig. 5.** Dependences of the increase in UTG power  $\Delta N_{utg}$ , reduction in fuel consumption  $\Delta G_{st}$ , reduces the specific fuel consumption  $\Delta g_e$  at different total pressure increase  $\pi_{c.atp}$  on the inlet air temperature  $t_{in}$ :

—————  $\pi_{c.atp} = 4.25$ ; - - - - -  $\pi_{c.atp} = 4.00$ ;  
 - · - · -  $\pi_{c.atp} = 3.85$ .

consumption by  $\Delta g_e = 4\text{--}7$  g/(kW·h), with a temporary reduction in fuel consumption  $\Delta G_{st} = 10\text{--}18$  kg/h (240–430 kg/day). The analysis of economic feasibility shows that the reduction in fuel consumption per voyage within 28 days will be  $\Delta G_v = 7\text{--}12$  tons. Taking into account the summer operation of the vessel, the reduction in fuel consumption per year will be  $\Delta G_{year} = 90\text{--}150$  tons.

## 5 Conclusion

The analysis of circuit solutions to use the thermopressor technologies for charge air cooling of engines is studied. It showed that the load reduction on the ship power plant with a corresponding reduction in the fuel consumption of diesel generators is 2–4%. In this case, the production of an additional amount of steam by 10–15% can be ensured.

The branches of the predominant application of thermopressor technologies include plants of stationary and marine power energy based on internal combustion engines and gas turbine engines. The use of thermopressor technologies provides a significant reduction in fuel consumption and can contribute to an increase in the energy security level of the state.

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